



STATISCHER PRÜFBERICHT FÜR CARPORT WING TYP 80

2023-09



STRUCTURE DESIGN ASSESSMENT REPORT

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1. Basic Information

1.1 Project Overview

TÜV Rheinland (hereafter refers to the **TR**) was entrusted by ETS Dienstleistungs- und Handels GmbH (hereafter refers to the **client**) as the third party to perform a structure design assessment regarding Carport Wing (hereafter refers to the **object**).

The structure design calculation and assessment are for the main body calculation of the object.

1.2 Principle Specifications

- 1) EN 1990:2002 Eurocode Basis of structural design;
- 2) EN 1991-1-1:2002 Eurocode 1: Actions on structures Part 1-1: General actions Densities, self-weight and imposed loads for buildings;
- 3) EN 1991-1-3:2003 Eurocode 1 Actions on structures Part 1-3: General actions Snow loads;
- 4) EN 1991-1-4:2005 Eurocode 1: Actions on structures Part 1-4: General actions Wind actions;
- 5) EN 1993-1-1:2005+A1:2014 Eurocode 3: Design of steel structures Part 1-1: General rules and rules for buildings;
- 6) EN 1999-1-1:2007+A1:2009+A2:2013 Eurocode 9: Design of aluminum structures Part 1-1: General structural rules;
- 7) 3D Model & Design description 2022.07;
- 8) Parts list & Drawings 2022.07.

1.3 Structure Overview

According to the data provided by client, the external contour dimension of the cantilevered shed carport structure is 4.70m (length) x2.40m (width) x2.10m (column height). The top of the object is an inclined structure with an inclination of 6°. The overall stressed frame structure of the object structure is composed of two 90.0mm x200.0mm x2.2/3.5mm rectangular pipe columns and two 68.0mm x 150.0mm x2.0mm cantilever beams. The columns and cantilever beams are connected by built-in corner pieces. Except that the connecting corner pieces are steel material, the other frame structure parts are all aluminum alloy material.

Component	Material	Width(mm)	Height(mm)	Thickness(mm)
Column	6082-T5	90.0	200.0	3.5/2.2
Cantilever beam	6082-T5	68.0	150.0	2.0
Top cross beam	6082-T5	38.0	46.0	1.0
Beam-column corner piece	Q235	-	-	4.0

Table 1-1 Dimensior	n parameter of	each component
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Figure 1-1 Overall design diagram of the object



Figure 1-3 Beam-column stiffening corner piece

Figure 1-2 Framework diagram of the object



Figure 1-4 Dimension drawing of stiffening corner piece

1.4 Material and Performance

The material mechanical parameters of steel structure framework of the object are shown as follows.

Materials	Elastic modulus (E) N/mm ²	Poisson's ratio (u)	Density (ρ) Kg/m³	Yield strength MPa
6082-T5	70,000	0.3	2,700	230
Steel material (Q235)	210,000	0.3	7,850	235

 Table 1-2 Material mechanical parameters

1.5 Load Value

For this object, the loads for consideration mainly include:

1) Wind load:

According to the information provided by client, the designed wind speed is 30.0m/s.

Referring to Clause 7.7 of EN 1991-1-4 (Wind load calculation method of rectangular section members), the value of wind load applied on the object is shown in Table 1-3~1-6.



Parameter name	Parameter	Value
Structural coefficient	CsCd	1
Roof angle (°)	α	6
Component force coefficient	Cf	0.7
Basic wind velocity (m/s)	Vb	30.0
Air density (kg/m ³)	ρ	1.25
Basic wind pressure (N/m ²)	$q_b = 0.5 \cdot \rho \cdot v_b^2$	562.5
Terrain category	-	I
Exposure factor	Ce	2
Peak wind pressure (N/m ²)	$q_p = c_e \cdot q_b$	1125.0
Reference area (m ²)	A _{ref}	11.2
Roof wind load (N)	$F_w = c_s \cdot c_d \cdot c_f \cdot q_p \cdot A_{ref} \cdot Sin(\alpha)$	920.3

 Table 1-3 Single inclined roof wind pressure load value

Table 1-4 Roof wind load friction load value

Parameter name	Parameter	Value
Structural coefficient	C _s C _d	1
Roof angle (°)	α	0
Friction coefficient	Cf	0.02
Basic wind velocity (m/s)	Vb	30.0
Air density (kg/m ³)	ρ	1.25
Basic wind pressure (N/m ²)	$q_b = 0.5 \cdot \rho \cdot v_b^2$	562.5
Terrain category	-	I
Exposure factor	Ce	2
Peak wind pressure (N/m ²)	$q_p = c_e \cdot q_b$	1125.0
Reference area (m ²)	A _{ref}	11.2
Roof wind load windward load (N)	$F_w = 2 \cdot c_s \cdot c_d \cdot c_f \cdot q_p \cdot A_{ref}$	504.0

Table 1-5 Y-direction each column wind load value

Parameter name	Parameter	Value
Section width (m)	b	0.09
Length (m)		2.10
Structural coefficient	CsCd	1
Component force coefficient	C _{f.0}	2.1
Section reduction factor	Ψr	1
Section reduction factor	Ψ_{λ}	1
Component force coefficient	$c_f = c_{f,0} \cdot \psi_r \cdot \psi_\lambda$	2.1
Basic wind velocity (m/s)	Vb	30.0
Air density (kg/m ³)	ρ	1.25
Basic wind pressure (N/m ²)	$q_b = 0.5 \cdot \rho \cdot v_b^2$	562.5
Terrain category	-	I
Exposure factor	Ce	2
Peak wind pressure (N/m ²)	$q_p = c_e \cdot q_b$	1125.0
Reference area (m ²)	A _{ref}	0.2
Wind load (N)	$F_w = c_s \cdot c_d \cdot c_f \cdot q_p \cdot A_{ref}$	446.5



Parameter name	Parameter	Value
Section width (m)	b	0.20
Length (m)		2.10
Structural coefficient	CsCd	1
Component force coefficient	C _{f.0}	2.1
Section reduction factor	Ψr	1
Section reduction factor	Ψ_{λ}	1
Component force coefficient	$c_f = c_{f.0} \cdot \psi_r \cdot \psi_\lambda$	2.1
Basic wind velocity (m/s)	Vb	30.0
Air density (kg/m ³)	ρ	1.25
Basic wind pressure (N/m ²)	$q_b = 0.5 \cdot \rho \cdot v_b^2$	562.5
Terrain category	-	I
Exposure factor	C _e	2
Peak wind pressure (N/m ²)	$q_p = c_e \cdot q_b$	1125.0
Reference area (m ²)	A _{ref}	0.4
Wind load (N)	$F_w = c_s \cdot c_d \cdot c_f \cdot q_p \cdot A_{ref}$	992.3

 Table 1-6 X-direction each column wind load value

2) Snow load:

According to the information provided by client, the snow load is 1.03kN/m².

3) Self weight and decorative plate:

According to the data provided by client, the self weight and decorative plate load shall be loaded by 1.5 times of the self weight of the framework.

1.6 Loading Conditions and Combinations

When calculating the structure of the object, the loads are combined according to Eurocode, and the main load combination conditions are as follows:

No	Namo		Tupo	Self weight of	Self weight of	Snow	Wind load	Wind load
INO	Name		туре	structure	roof	load	WX	WY
1	gLCB1	Active	Added	1.35	1.35	1.5		
2	gLCB2	Active	Added	1.35	1.35		1.5	
3	gLCB3	Active	Added	1.35	1.35			1.5
4	gLCB4	Active	Added	1.35	1.35		-1.5	
5	gLCB5	Active	Added	1.35	1.35			-1.5
6	gLCB6	Active	Added	1.35	1.35	1.05	1.5	
7	gLCB7	Active	Added	1.35	1.35	1.05		1.5
8	gLCB8	Active	Added	1.35	1.35	1.05	-1.5	
9	gLCB9	Active	Added	1.35	1.35	1.05		-1.5
10	gLCB10	Active	Added	1.35	1.35	1.5	0.9	
11	gLCB11	Active	Added	1.35	1.35	1.5		0.9
12	gLCB12	Active	Added	1.35	1.35	1.5	-0.9	
13	gLCB13	Active	Added	1.35	1.35	1.5		-0.9

Table 1-7 Loading conditions and combinations



2. Establishment of Model

Because of the complexity of such object structure, it will be inefficient to perform the calculation by integral entity finite element calculation, so the calculation analysis is performed with the finite element software - Midas GEN.

2.1 Unit Selection

During overall structure analysis, each member of the structure is simulated by beam elements.

2.2 Boundary and Constraints

Since the column of the final assembled structure is embedded into the ground, and the embedded depth is 0.5m, so the lower part of the column is calculated as the fixed connection.





Figure 2-1 Overall structural framework diagram

Figure 2-2 Connection diagram of cantilever and cross beam



Figure 2-3 Diagram of column bottom fixation



3. Stress Analysis of Integral Structure

As the framework of the object structure is the main stress component, the beam element is mainly used to simulate the stress of the framework during stress analysis of the structure.

3.1 Introduction of Finite Element Model and Load Application of Structure

During the service of the object, external live loads mainly include wind load and snow load. Because the object may be subjected to wind loads from all directions, which can be divided into X-direction wind load and Y-direction wind load. The application methods of wind load and snow load are shown in Figure 3-1~3-3.





3.2 Stress Analysis under Ultimate Load Bearing Capacity Condition at Service Stage

When analyzing the structure, the stress condition of the structure can be divided into ultimate load bearing capacity condition and ultimate normal service condition, which correspond to the displacement control of the structure and the stress control of structural materials respectively.

This structure is a single-layer assembled structure. So under ultimate normal service condition, when the displacement of the structure is large, it will not affect the service of the structure. Therefore, for this object structure, only analysis calculation of ultimate load bearing capacity



condition is carried out.

In order to analyse the most unfavorable stress condition of structure under ultimate load bearing capacity conditions, this section analyses the stress envelope diagram of the structure under ultimate load bearing capacity conditions, and also the stress of the structure under individual load bearing capacity conditions.

During the analysis of the structural framework, since the overlapping length of the corner piece and the cantilever beam and the column is close to 0.5m, the load at the connection between the column and the cantilever beam is born by the stiffening corner piece under the actual load, so the range of 0.4m at the connection position between the column and the cantilever beam is not considered for analysing.

Stress envelope of structure under each ultimate load bearing capacity condition is shown in Figure 3-4. According to the analysis of the envelope diagram, under ultimate load bearing capacity condition, maximum compressive stress of the structure is 202.8MPa, maximum tensile stress is 201.4MPa. Both stresses are lower than material yield strength of 230.0MPa. So material yield failure will not happen, which meets the design requirements of code.

Stress envelope of column under each ultimate load bearing capacity condition is shown in Figure 3-5. According to the analysis of the envelope diagram, under ultimate load bearing capacity condition, maximum compressive stress of the column is 164.8MPa, maximum tensile stress is 151.6MPa. Both stresses are lower than material yield strength of 230.0MPa. So material yield failure will not happen, which meets the design requirements of code.

Stress envelopes of cantilever beam under each ultimate load bearing capacity condition are shown in Figure 3-6. According to the analysis of the envelope diagram, under ultimate load bearing capacity condition, maximum compressive stress of cantilever beam is 202.8MPa, maximum tensile stress is 201.4MPa. Both stresses are lower than material yield strength of 230.0MPa. So material yield failure will not happen, which meets the design requirements of code.

Stress envelopes of cross beam under each ultimate load bearing capacity condition are shown in Figure 3-7. According to the analysis of the envelope diagram, under ultimate load bearing capacity condition, maximum compressive stress of cross beam is 167.7MPa, maximum tensile stress is 167.7MPa. Both stresses are lower than material yield strength of 230.0MPa. So material yield failure will not happen, which meets the design requirements of code.

Structural stress diagram under each load bearing capacity condition of gLCB1 ~ gLCB13 is shown in Figure 3-8~3-20. And Table 3-1 shows the statistics of maximum compressive and tensile stress of structure under each ultimate load bearing capacity condition of gLCB1 ~ gLCB13. According to these statistical data, the most unfavorable load conditions of structure are gLCB10 and gLCB12, and load combination of stress is 1.35 (constant load) + 1.50 (snow load) \pm 0.90WX (wind load). Therefore, the most unfavorable stress of the structure is the under these two load conditions.

In summary, under ultimate load bearing capacity condition, material yield failure will not happen at



the column (maximum stress 164.8MPa), cantilever beam (maximum stress 202.8MPa) and cross beam (maximum stress 167.7MPa) of frame structure. So the structure under ultimate load bearing capacity condition meets the design requirements of code.

No	Name	Beam & Column		
		Maximum compressive stress	Minimum tensile stress	
		MPa	MPa	
1	gLCB1	195.8	159.5	
2	gLCB2	62.0	56.4	
3	gLCB3	16.7	11.3	
4	gLCB4	62.0	5.6	
5	gLCB5	34.5	15.2	
6	gLCB6	152.7	113.3	
7	gLCB7	140.2	122.4	
8	gLCB8	152.7	113.3	
9	gLCB9	122.4	142.0	
10	gLCB10	202.8	159.5	
11	gLCB11	159.3	167.7	
12	gLCB12	202.8	159.5	
13	gLCB13	196.4	167.7	

Table 3-1 Maximum stresses of structure under each load bearing capacity conditions













Figure 3-7 Stress envelope diagram of cross beam under ultimate load bearing capacity condition (-167.7 ~ 167.7MPa)



Figure 3-8 Structural stress under gLCB1 load condition (-195.8 ~ 159.5MPa)







Figure 3-10 Structural stress under gLCB3 load condition (-16.7 ~ 11.3MPa)



Figure 3-11 Structural stress under gLCB4 load condition (-62.0 ~ 5.6MPa)



Figure 3-12 Structural stress under gLCB5 load condition (-34.5 ~ 15.2MPa)



Figure 3-13 Structural stress under gLCB6 load condition (-152.7 ~ 113.3MPa)



Figure 3-14 Structural stress under gLCB7 load condition (-140.2 ~ 122.4MPa)



Figure 3-15 Structural stress under gLCB8 load condition (-152.7 ~ 113.3MPa)



Figure 3-16 Structural stress under gLCB9 load condition (-122.4 ~ 142.0MPa)



Figure 3-17 Structural stress under gLCB10 load condition (-202.8 ~ 159.5MPa)



Figure 3-20 Structural stress under gLCB13 load condition (-196.4 ~ 167.7MPa)

3.3 Chapter Summary

After overall calculation and analysis on the object structure by Midas GEN, main results and conclusions are shown as follows:

- 1) Under ultimate load bearing capacity condition, maximum compressive stress of the structure with member system unit is 202.8MPa, maximum tensile stress is 201.4MPa, and both stresses are lower material yield strength of 230.0MPa. So material yield failure will not happen, which meets the design requirements of code.
- Under ultimate load bearing capacity condition, the most unfavorable load combination conditions of structure are gLCB10 and gLCB12. The corresponding load combination is 1.35 (constant load) + 1.50 (snow load) ± 0.90WX (wind load).



3) Under ultimate load bearing capacity condition, material yield failure will not happen at the column (maximum stress 164.8MPa), cantilever beam (maximum stress 202.8MPa) and cross beam (maximum stress 167.7MPa) of frame structure. So the structure under ultimate load bearing capacity condition meets the design requirements of code.



4. Stress Analysis of Local Structure

For this structure, the local area with complicated stress is the beam-column connecting corner piece, whose main stress is to transfer the bending moment at the root of the cantilever beam to the column. Therefore, in this section, the local stress of the connecting corner piece is analyzed by solid finite element.

4.1 Finite Element Model and Load Application of Corner Piece Structure





Figure 4-1 Bending moment envelope of corner piece under ultimate load bearing capacity condition (11280Nm)



Figure 4-2 Diagram of corner piece finite element model



4.2 Calculation Results of Corner Piece Structure

According to the calculation results of Figure 4-5~4-7, under ultimate load bearing capacity condition, the maximum principal tensile stress of the corner piece is 221.6MPa, the maximum principal compressive stress is 261.7MPa, and the maximum Mises stress is 229.3MPa. The maximum principal compressive stress is higher than the material yield strength of 235.0MPa. So the corner piece will yield locally, and the yield position is at the lower angle of the corner piece. According to Figure 4-8, the maximum plastic strain at the stress concentration area of the corner piece is 0.0024, which is lower than the allowable material value of 0.1, so failure will not happen at the corner piece.

In summary, under ultimate load bearing capacity condition, failure will not happen at the corner piece, and the corner pieces structure meets the design requirements of code.







Figure 4-4 Diagram of corner piece diaplacement (3.26mm)



A: BCASH PU-5207 HC M/H BC M/H BC

Figure 4-7 Principal compressive stress (261.7MPa)



Figure 4-8 Diagram of equivalent plastic strain (0.0024)

Figure 4-5 Mises stress (229.3MPa)



5. Conclusions

Calculation results of overall and local structure framework by using Midas GEN and ANSYS software are shown as follows:

- 1) Under ultimate load bearing capacity condition, maximum compressive stress of the structure with member system unit is 202.8MPa, maximum tensile stress is 201.4MPa, and both stresses are lower material yield strength of 230.0MPa. So material yield failure will not happen, which meets the design requirements of code.
- Under ultimate load bearing capacity condition, the most unfavorable load combination conditions of structure are gLCB10 and gLCB12. The corresponding load combination is 1.35 (constant load) + 1.50 (snow load) ± 0.90WX (wind load).
- 3) Under ultimate load bearing capacity condition, material yield failure will not happen at the column (maximum stress 164.8MPa), cantilever beam (maximum stress 202.8MPa) and cross beam (maximum stress 167.7MPa) of frame structure. So the structure under ultimate load bearing capacity condition meets the design requirements of code.
- 4) Through the local calculation of corner piece structure by solid finite element method, under ultimate load bearing capacity condition, failure will not happen at the corner piece, and the corner pieces structure meets the design requirements of code.

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